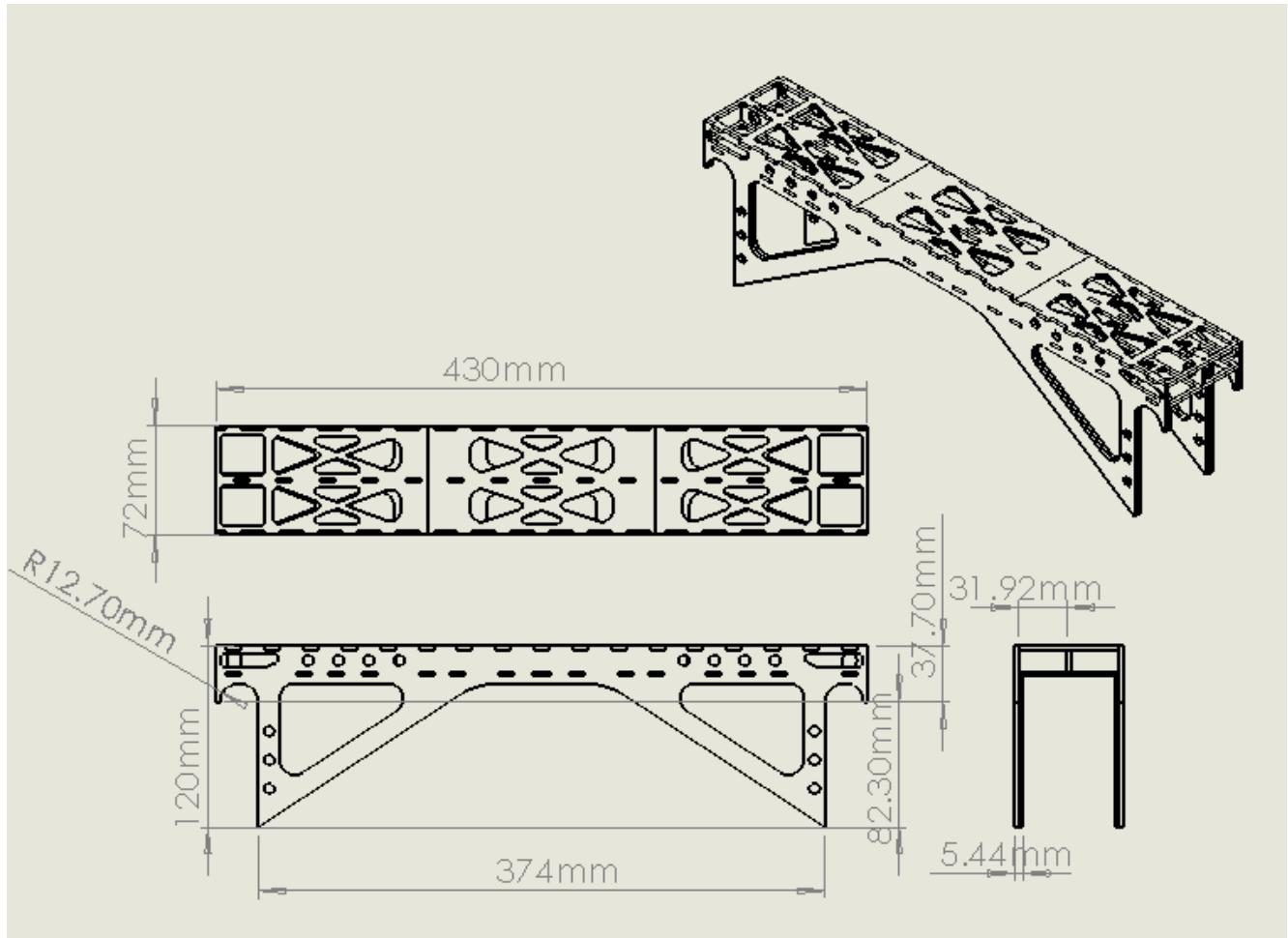


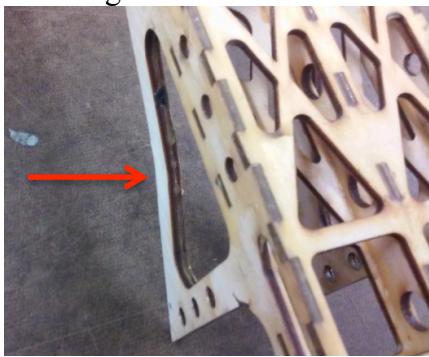
Stephanie Emenyonu, Callan George, William Hickman, Matthew McFarland and Tatjana
Toeldte
Group 1
ENGS 033 Final Bridge Report

Final Tested Design:



SolidWorks

Predicted Results vs. Actual Results:

	Deflection under 1kN	Critical Load	Location of Failure
Predicted Results	0.1 mm	1.7 kN	
Actual Results	1.678 mm	3.384 kN	Buckling of side truss members 

Analytical Results: See attached.

Post-testing Discussion:

During testing, the bridge failed when the angled side truss members buckled under the critical load. It failed there because the strength of the I-beam deck combined with the support from the jig on the vertical members of the side trusses left the angled truss members more vulnerable to buckling.

Our predicted deflection under 1 kN of force was approximately 6% of the actual deflection under 1 kN. Also, our critical load prediction was approximately 50.2% of the actual critical load applied before failure. However, our bridge did relatively well concerning the failure load to weight ratio and bridge size: The failure to weight ratio was approximately 17.66 N/gram, the highest ratio of all the groups, and the bridge weight was about 191.6 grams, the smallest of all the groups. This was beneficial for our bridge because optimal values for these two characteristics are critical in bridge construction and material management. In addition, our bridge performed relatively well in the deflection*weight category with 321.5 mm*grams.

To improve the accuracy of the analysis, we could have been less conservative with our Phase 3 calculations. We originally had a predicted value of about 3.0 kN and also had a SolidWorks predicted value of 3.6 kN as the critical load; however, we decided to take into consideration imperfection factors such as gluing and bridge assembly, causing us to decrease our predicted value by about a factor of 2 to 1.7 kN. Without consideration of those factors, our

load prediction would have been more accurate at about 88.7% of the actual critical load. The deflection prediction could have been improved by considering that SolidWorks does not take buckling into account significantly during model analysis and that SolidWorks also takes into account warping of the model that is realistically unfeasible when calculating deflection.

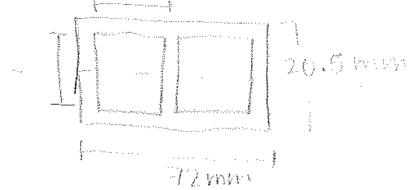
Realizing this, we could have increased our prediction value for deflection.

In order to improve the scale-bridge model, we could increase the strength in the side trusses that had buckled during testing by adding a cross-brace between adjacent trusses on the shorter sides of the assembly in order to limit buckling. We could also reinforce the deck with a smaller upper truss or a more elaborate lower truss to limit buckling in the lower truss region.

Analytical Calculations

Note About prediction ①

Deck simplified as 60% beam:



$$I = \frac{1}{12} (0.072 \times 0.0205 - 2 \times \quad \times \quad)$$

$\approx 3.4 \times 10^{-8} \text{ m}^4$

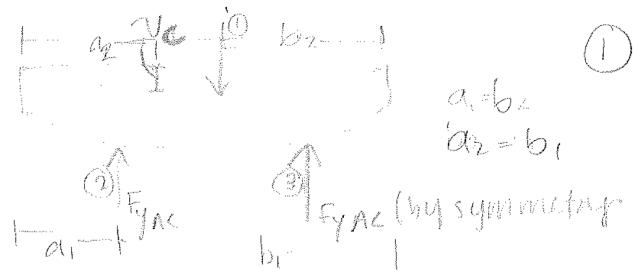
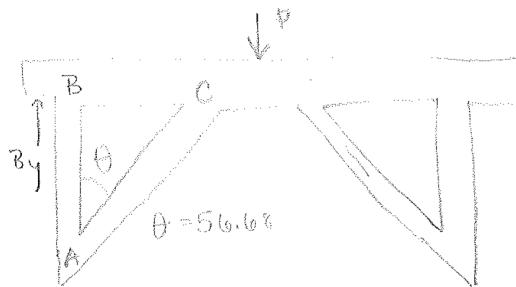
70 MPa \Rightarrow compression 70% as good \Rightarrow 50 MPa

lowest value 3.0 kN

\hookrightarrow to account for bridge imperfections \rightarrow

use factor of 2

$$\frac{3.0}{2} = \left\{ \begin{array}{l} 1.7 \text{ kN is the} \\ \text{critical load} \end{array} \right\}$$



$V_c = S_{AC}$ due to F_{AC} developed

POINT FORCE @
center

$$\frac{F_{AC}L}{A\Delta} = \frac{-Pa}{48EI} (3L^2 - 4a^2) \quad (1) \quad + F_{AC} \cos(56.68^\circ) b_1 a_1 (L^2 - b_1^2 - a_1^2) \quad (2)$$

↓

$$+ \frac{F_{AC} \cos(56.68^\circ) b_2 a_1 (L^2 - b_2^2 - a_1^2)}{6EI} \quad (3)$$



$$48FACL^2I = -PAa_1L(3L^2-4a^2) + ABFAC(0.5L^2a^2) + [b_1(L^2-b_1^2-a_1^2) + b_2(L^2-b_2^2-a_2^2)]$$



$$FAC \left[48L^2 I - 84605(56.68) a_1 [\sim] \right] = - P A a_1 L (3L^2 - 4a^2)$$



$$= -PAA_1L(3L^2-4a_1^2) \\ \frac{48L^2T - 8A\cos(\theta_0, b_0)a_1[b_1(L^2-b_1^2-a_1^2) + b_2(L^2-b_2^2-a_1^2)]}{48L^2T - 8A\cos(\theta_0, b_0)a_1[b_1(L^2-b_1^2-a_1^2) + b_2(L^2-b_2^2-a_1^2)]}$$

$$q_1 = +1100m$$

$$b_1 = 1.246 \text{ m}$$

$$b_2 = 1160 \text{ m}$$

$$L = 406 \text{ m}$$

$$T = 3.4 \times 10^{-8} \text{ m}^{-4}$$

$$A = 2 \times 0.00272 \times 0.012 \text{ m}^2 \\ = 6.528 \times 10^{-5}$$

$$F_{AC} = -P(6.538E-5)(.16)(.406) \left[3(.406)^2 - 4(.16)^2 \right] \\ \frac{48(.406^2)(3.4E-8) - 8(6.538E-5)5.60653.68)(.16) \left[(2.406)(.406^2 - .2406^2 - .16^2) \right. \\ \left. + (.16)(.406^2 - .16^2 - .16^2) \right]$$

$$F_{AC} = \frac{-P(1.663 \times 10^{-6})}{2.69 \times 10^{-7} - 1.735 \times 10^{-6}} = F_{AC} = 7.17 \text{ (P)}$$

+ RANKINE

$F_{ac} = 702 \text{ Pa} \rightarrow \text{compression}$

1/8" cross grain in
tension, $\sigma_{max} \sim 70 \text{ MPa}$

tension, $\sigma_{\max} \sim 70 \text{ MPa}$

wood performs worse in compression.

Let's use $\sigma_{\text{max}} \sim 45 \text{ MPa}$ (60%)

$$45 \text{ Mba} = 1.7 \cdot P$$

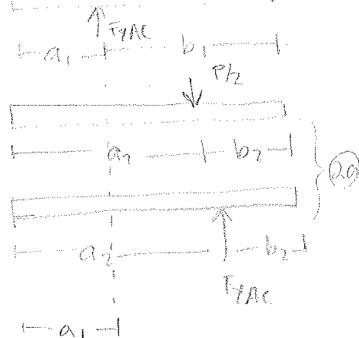
$$45(10^6)(6.525 - 5) = 1.7 \text{ MN}$$



| 2 LOADS 150 MM APART |

$$\textcircled{1} \quad P_1 \downarrow \quad P_2 \downarrow \quad \delta_{AC} = \nu_c \quad \textcircled{10}$$

$$\textcircled{1} \quad P_1 \downarrow \quad \left. \begin{array}{l} \textcircled{10} \\ \textcircled{11} \end{array} \right\} \quad \frac{F_{AC} L}{A E} = \frac{(F_{AC} \cos(56.68) - \frac{P}{2})(b_1 a_1)(L_2 - b_1^2 - a_1^2)}{6 \nu I L} + \frac{(F_{AC} \cos(56.68) - \frac{P}{2})(b_2 a_2)(L_2 - b_2^2 - a_2^2)}{6 \nu I L}$$



$$6F_A L^2 I = A F_{AC} \cos(56.68)(\textcircled{10} + \textcircled{11}) - \frac{AP}{2}(\textcircled{10} + \textcircled{11})$$

$$F_{AC} (6L^2 I - A \cos(56.68)(\textcircled{10} + \textcircled{11})) = -\frac{AP}{2}(\textcircled{10} + \textcircled{11})$$

$$F_{AC} = \frac{-AP(\textcircled{10} + \textcircled{11})}{6L^2 I - A \cos(56.68)(\textcircled{10} + \textcircled{11})}$$

$$a_1 = .160 \text{ m}$$

$$b_1 = .246 \text{ m}$$

$$b_2 = .160 \text{ m}$$

$$L = .406 \text{ m}$$

$$I = 3.4 \times 10^{-8} \text{ m}^4$$

$$A = 6.528 \times 10^{-5} \text{ m}^2$$

$$\textcircled{10} = (.246)(.160)(.406^2 - .246^2 - .160^2) = 0.0036984$$

$$\textcircled{11} = (.160)(.160)(.406^2 - .160^2 - .160^2) = 0.00290908$$

$$F_{AC} = \frac{-P(3.92E-7)}{3.36E-8 - 2.145E-7} = 2.17 P$$

$$\delta_{max} = \frac{2.17 P}{6.528 \times 10^{-5}}$$

σ_{max}	P_{max}
70 MPa	2.1 kN
60 MPa	1.8 kN
50 MPa	1.5 kN
40 MPa	1.2 kN

used $\frac{1}{8}$ " data because peg structure for

$\textcircled{10}$ is different from $\textcircled{11}$

$\frac{1}{8}$ " $\frac{1}{4}$ "

but, with the glue present, it's probably
between the two (in tension)

actually, load is distributed
over 2 masses

$$\frac{P_{max}}{2} = 1.05 \text{ kN}$$

70	1.2	solidworks material value
60	3.6	
50	3.0	
40	2.4	

Predictions

Deflection with 1kN

$$0.1 \text{ mm}$$

Critical Load

With imperfections (1.7 kN)

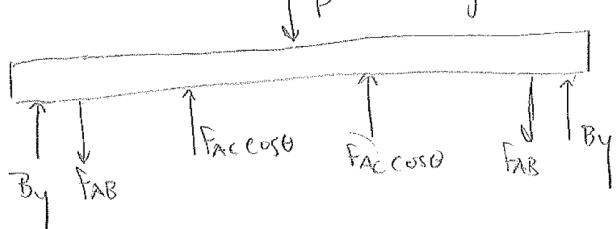
$$\text{raw } 3.0 \text{ kN} \rightarrow 3.6 \text{ kN}$$

Using the value for F_{AC} found on page 1, a point analysis of A reveals that $F_{AB} = F_{AC} \cos(56.68^\circ)$ $F_{AC} = 1.77$ (3)

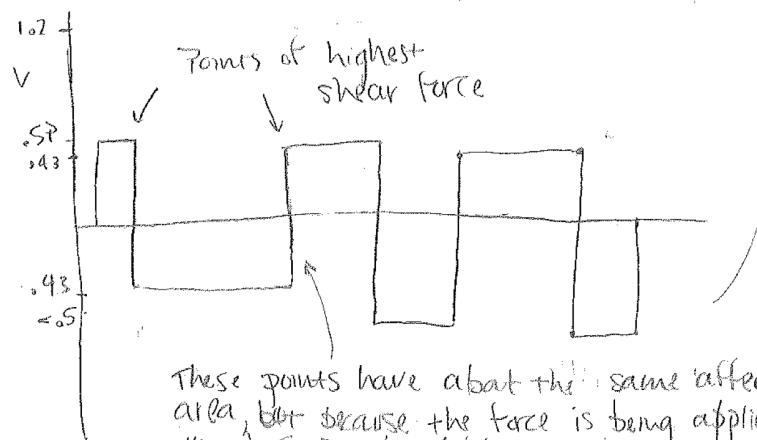
$$\begin{array}{l} \text{At } A: \sum F_y = 0 \Rightarrow F_{AB} - F_{AC} \cos(\theta) = 0 \\ \theta = 56.68^\circ \\ F_{AB} = F_{AC} \cos(56.68^\circ) = 93 \text{ N} \end{array}$$

A quick analysis of external forces indicates that $By = \frac{P}{2}$ and that the two sides of the bridge are symmetrical.

Now to consider the bridge deck:



Shear force diagram:



These points have about the same effective area, but because the force is being applied closer to this point, deflection affects it more, meaning it's more likely to break there first.

$$T_{\max} = 1.5 \frac{V}{A} \quad A = (6107)(0.544) \\ A = 5.8268 \times 10^{-5} \text{ m}$$

$T_{\max} = 12884,83\text{ P}$

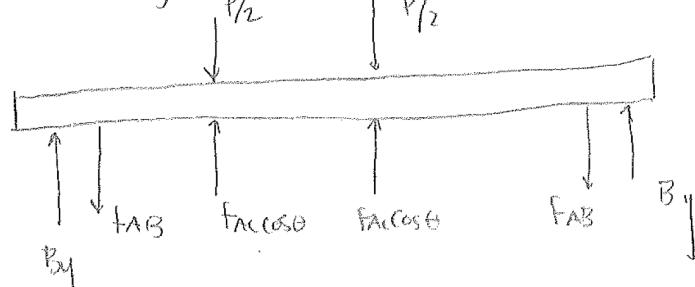
$$\begin{aligned}
 70 \text{ MPa} &\rightarrow 5.4 \text{ KN} \\
 60 \text{ MPa} &\rightarrow 4.6 \text{ KN} \\
 50 \text{ MPa} &\rightarrow 3.9 \text{ KN} \\
 45 \text{ MPa} &\rightarrow 3.4 \text{ KN}
 \end{aligned}$$

Analysis for point force at Center
* Bridge Deck

Using the values found on page 2, and the point analysis of A on page 3 $F_{AB} = F_{AC} \cos \theta = 1.192 P$:

4

External force analysis reveals that By still equals $P/2$.
 Considering the bridge deck:



$$T_{max} = 1.5 \frac{V}{A} \quad A = 5.8208 \times 10^{-5} \text{ m}$$

$$T_{max} = \frac{(1.5)(-692)}{A} P = 17832.6 \text{ P}$$

max range \rightarrow Breaking force

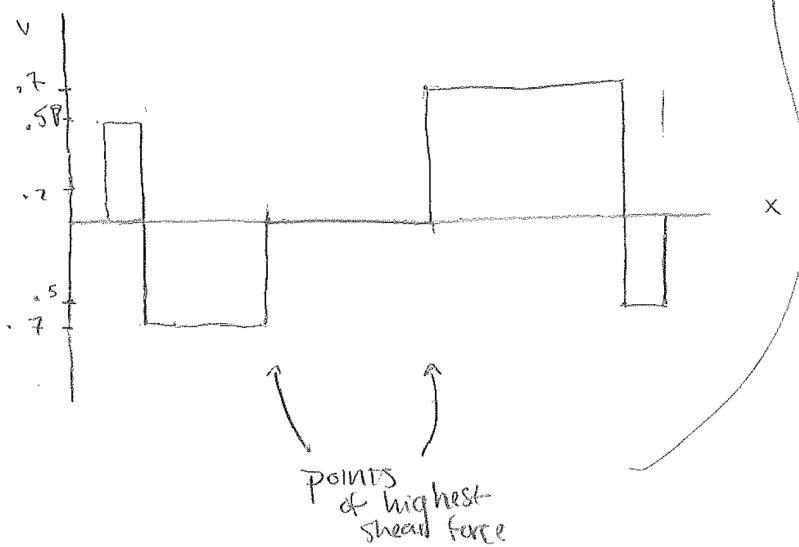
$$70 \text{ MPa} \rightarrow 3.9 \text{ kN}$$

$$60 \text{ MPa} \rightarrow 3.36 \text{ kN}$$

$$50 \text{ MPa} \Rightarrow 2.8 \text{ KN}$$

$$45 \text{ MPa} \rightarrow 2.5 \text{ KN}$$

Shear force diagram



points of highest shear force

Analysis for two point forces 150 mm apart
* Bridgedeck